



SERVICE LETTER

No. 676

Piper Aircraft Corporation

Lock Haven, Pennsylvania, U.S.A.

October 4, 1973

M

Subject:

Weight and Balance/Equipment List Correction

Models Affected:

PA-23-250 (Six Place) and
PA-E23-250 (Six Place) Aztec
-- 5200 lb. gross weight only

Serial Numbers Affected:

27-2505 to 27-3836 incl., 27-3838 to 27-3943 incl. --
5200 lb. gross weight only

Compliance Time:

At the aircraft owner's/operator's discretion.

Purpose:

1. Applies to PA-23-250 (Six Place) Aztec models;
to provide proper information to correct an erroneous
entry in the Equipment List, Page 14, Section II,
Report No. 1360.
2. Applies to PA-E23-250 (Six Place) Aztec models;
to provide revised Equipment List page for Report
No. 1378 for the reason stated above.

Although this situation is of a non-serious nature, it is felt
that the correction should be made to preclude confusion
that may exist as a result of an incorrect Equipment List
reference.

Instructions:

1. Reference Purpose No. 1, above;
 - a. Refer to Report No. 1360, Weight and Balance/
Equipment List, Page 14, Section II.
 - b. Under Landing Gear, refer to the following entry:

"Cleveland
Wheel Assembly Model 3080C
Brake Assembly Model 37-72"
 - c. Delete the "37-72" brake assembly model design-
ation and type in "37-200A". The corrected
entry should read as follows:

Cleveland
Wheel Assembly Model 3080C
Brake Assembly 37-200A
2. Refer to Purpose No. 2, above;
 - a. Refer to Report No. 1378, Weight and Balance/
Equipment List, Page 14, Section II. (over)

Instructions: (continued)

2. b. Replace existing page with revised page dated 8/27/73.
3. Make appropriate log book entry.

Material Required:

PA-E23-250 (Six Place) Aztec -- 5200 lb. gross weight models; one each per aircraft Report No. 1378, revised Weight and Balance/Equipment List page 14, Section II -- revision date 8/27/73.

Availability of Parts:

Your Piper Dealer.

Effectivity Date:

This Service Letter is effective October 15, 1973.

Summary:

Compliance with the provisions of this Service Letter may be achieved by the affected aircraft owner/operator in accordance with Instructions, above.